

GNSS for Ionospheric Studies

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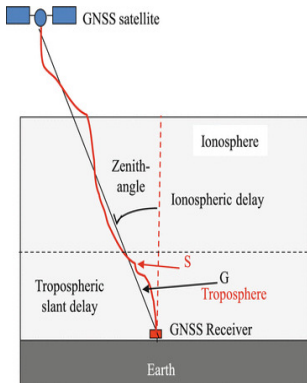
Global Navigation Satellite Systems (GNSS)

A satellite system primarily for navigation. Now days GNSS is used for surveying, geodesy, geophysics and meteorology.

GNSS include

- Global Positioning System (GPS) - USA
- GLObal NAVigation Satellite System (GLONASS)
- Galileo - Europe
- Compass or (Beidou-2) - China
- etc

GNSS signal propagation



- The atmosphere delays Radio signals
- The excess path \propto refractive index (N)

- Total delay = $\underbrace{\int_R^S N_A(\rho) dl}_{\text{Tropospheric}} + \underbrace{\int_R^S N_I(n_e) dl}_{\text{Ionospheric}}$

- Tropospheric = $\underbrace{\text{ZHD}}_{\text{dry delay}} + \underbrace{\text{ZWD}}_{\text{wet delay}}$

- The ionospheric delay is given by

$$I = \pm \frac{40.3}{f^2} \int_R^T N_e ds$$

How is the TEC derived?

- GNSS signals are transmitted at two coherent frequencies
- The delays on each frequency are

$$P1 = \rho + I_{1g} + \gamma_1^r + \gamma_1^s \quad (a)$$

$$P2 = \rho + I_{2g} + \gamma_2^r + \gamma_2^s \quad (b)$$

$$L1 = \rho + I_{1\phi} + \Omega_1 + \varepsilon_1^r + \varepsilon_1^s \quad (c)$$

$$L2 = \rho + I_{2\phi} + \Omega_2 + \varepsilon_2^r + \varepsilon_2^s \quad (d)$$

- The delays are computed by the receiver

A typical obseravation file

```
1      2.11      OBSERVATION DATA      G (GPS)      RINEX VERSION / TYPE
2 teqc 2010Mar17      gpsops      20110414 21:54:45UTC PGM / RUN BY / DATE
3      30.0000      INTERVAL
4      1      1      WAVELENGTH FACT L1/2
5      7      L1      L2      P1      P2      C1      S1      S2      # / TYPES OF OBSERV
6 Linux 2.4.21-27.ELsmp|Opteron|gcc -static|Linux x86_64|=+      COMMENT
7      2.10      OBSERVATION DATA      GPS      COMMENT
8 soc2rnx ver 1.20      gpsops      14-Apr-2011 21:54:45      COMMENT
9 S1, if present, is the SNR for the C/A data stream on L1.      COMMENT
10 SNR is mapped to RINEX snr flag value [1,4-9]      COMMENT
11      SNR: >316 >100 >31.6 >10 >3.2 >0 bad=0      COMMENT
12 L1 & L2:      9      8      7      6      5      4      1      COMMENT
13 GGN      JPL      OBSERVER / AGENCY
14 CR519993001      ASH701945B_M      SCIS      ANT # / TYPE
15 5482951.3876 3260442.6429 -66519.8489      APPROX POSITION XYZ
16 0.0083 0.0000 0.0000      ANTENNA: DELTA H/E/N
17 UC120031617      ASHTECH UZ-12      CQ00      REC # / TYPE / VERS
18 MBAR      MARKER NAME
19 33901M001      MARKER NUMBER
20      COMMENT
21 This data is provided as a public service by NASA/JPL.      COMMENT
22 No warranty is expressed or implied regarding suitability      COMMENT
23 for use. For further information, contact:      COMMENT
24 Dave Stowers, NASA/JPL m/s 238-600      COMMENT
25 4800 Oak Grove Drive, Pasadena CA 91109 USA      COMMENT
26      COMMENT
27 Forced Modulo Decimation to 30 seconds      COMMENT
28 2011 4 9 0 0 0.0000000 GPS      TIME OF FIRST OBS
29      END OF HEADER
30 11 4 9 0 0 0.0000000 0 7G25G31G29G22G21G18G12
31 108500900.84747 84546128.03247 20647055.3234 20647062.0414 20647056.1884
32 50.000 45.000
33 122288475.36247 95289692.80647 23270744.9294 23270750.9654 23270743.8644
34 40.000 36.000
35 124235097.42547 96806562.51647 23641161.1934 23641166.5824 23641161.1134
36 40.000 35.000
37 120216270.87147 93675004.93347 22876404.9704 22876406.3404 22876403.3104
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How is the TEC derived?

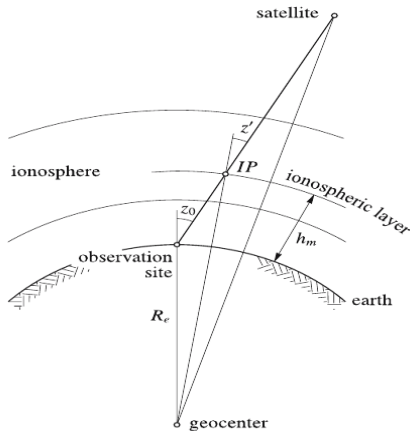
- TEC is obtained from the geometry-free linear combination

$$(P2 - P1) = 40.3s\text{TEC}_g \left(\frac{1}{f2^2} - \frac{1}{f1^2} \right) + \text{BRS}_g$$

$$(L1 - L2) = 40.3s\text{TEC}_\phi \left(\frac{1}{f2^2} - \frac{1}{f1^2} \right) + (\Omega_1 - \Omega_2) + \text{BRS}_\phi,$$

- TEC obtained from the above equations is slant TEC

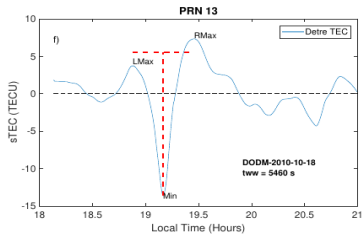
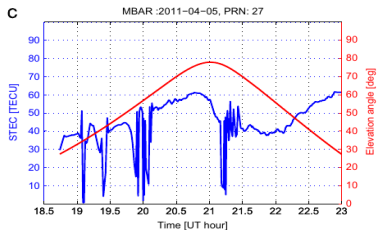
Vertical TEC



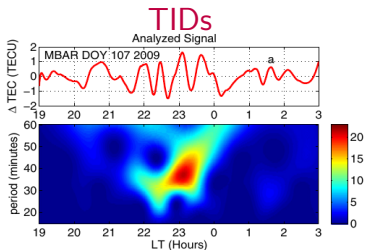
- sTEC is often converted to vTEC using thin shell model
- $\sin z' = \frac{R_e}{R_e + h_m} \sin z_0$,
- The sTEC and the vTEC related by

$$\text{sTEC} = \frac{1}{\cos z'} \text{vTEC}$$

What information can be obtained from TEC?



EPBs



TIDs

EPBs

THANK YOU